

# A-Class Catamaran Board Hydrodynamics

Tom Speer http://www.tspeer.com/Aclass/A-ClassCatamaranFoils.pdf

# **Questions Considered**

When an A-Class Catamaran is flying on foils,

- How does board design contribute to flying stability?
- What daggerboard geometries fit the A-Class Rule?
- How do board characteristics change with curvature?
- What is the tradeoff between heave stability & drag?
- Where might A-Class board design go in the future?

Focus is on understanding, not optimization

Lift-induced drag only – no profile drag, no wave drag



#### Terms



## Requirements



## **Equilibrium Forces & Moments**



∑Vertical Lift = Total Weight

 $\Sigma$ Horizontal Lift = Side Force

Side Force \* Center of Effort = Hydrodynamic Moment + Sailor Weight \* Lever Arm

Leeway & angle of attack adjust to ensure equilibrium

Lifting line analysis
 considers wake shape
 in plane behind boards

# Static Stability Requirements

- Assume sway, roll & yaw axes in equilibrium
- Start from trimmed condition
- Heave stability: dFz/dh < 0
   <p>An increase in flying height
   at constant pitch attitude => reduction in lift
- Pitch stability: dMy/dθ < 0 An increase in pitch attitude at constant flying height => bow-down moment
- Pitch-heave coupling: dMy/dh < 0
   <p>An increase in flying height
   at constant pitch attitude => bow-down moment

# **Design for Static Stability**

#### Heave stability

- Surface piercing foils (V, ladder)
- Active feedback control (Moth, Rave, Trifoiler)
- Leeway-modulated lift (AC72)
- Pitch stability
  - Aft foil less heavily loaded than forward foil (per m<sup>2</sup>)
  - Weight forward

#### Pitch-heave coupling

- Forward foil has higher heave stiffness than aft foil
- Fully submerged aft foil (T rudders)
- Stern-first takeoff
- This talk only concerns heave stability of boards

# **A-Class Design Requirements**

- Vertical lift = 160 kg
  - 75 kg boat
  - 85 kg crew
- Righting moment (about centerline) = 180 kg m
- Height of center of effort = 2.5 m (assumed)
- A-Class span limits
  - Beam < 2.3 m
  - Foils > 0.75m from centerline
- Exit from hull ~ 1.0 m from centerline
  - Demihull beam <0.3m</li>
- Rigid boards are assumed
- Port-starboard symmetry

# **Design Parameters**

- Geometry
  - Trunk location
  - Center depth
  - Bend radius
  - Wing dihedral
  - Board chord
  - Wing tip chord
- Operating Conditions
  - Flying height (h)
  - Heel angle (0, 10)
  - Speed (15 kt)





## On to Foils!



#### **Shapes Considered**



## C Board Load Distributions



#### **C** Board Characteristics



# C Board Stability & Drag

- Leeway increases as boat flies higher
- Stability initially decreases as boat flies higher





#### Drag increases as boat flies higher

• Lift and drag shift to leeward board

#### **Shapes Considered**



## **Board Drag Comparisons**



## **Board Stability Comparisons**



## **Righting Moment Comparisons**



#### Side Force Comparisons



## C Board Loading



## Effect of Toe-In/Out



## Effect of Toe-In/Out



## Effect of Toe-In/Out





## Effect of Twist at Top



## Effect of Twist at Bottom



## Effect of Twist



## Heave Stability vs Drag Tradeoff



## What the Elephant Looks Like So Far

- C, J, L, < shapes investigated with lifting line
  - Allowable sail power is significantly reduced
  - Twin foils + symmetry = loss of righting moment
- Anhedral improves stability, adds drag (both boards in water)
- Heeling to leeward is destabilizing but adds sail power
- For C boards:
  - Toe-in is destabilizing
    - No effect on drag or sail power
  - Toe-out is stabilizing
    - Increases sail power, but large drag penalty
  - Wash-out reduces drag of rectangular planform
    - No effect of twist on sail power
    - Stability improved

## Ideas For The Future



# Problem of Symmetry



- Angle of attack affects both boards equally
- Equal vertical lift has no righting moment
- Need to rake boards differentially

#### L Foil With Anhedral



## Effect of Differential Rake



# **Floating Flap**

- Flap hinge moment is function of:
  - Flap shape (trailing edge camber)
  - Flap deflection
  - Angle of attack
- Flap free => hinge moment = 0
- Negative flap deflection reduces lift



## NACA 63012a With 30% Chord Flap



## Fine

